

Instrument Synthesis and Analysis Laboratory

# Orbiting Wide-angle Light-collectors (OWL)

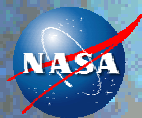
**Mechanical Subsystem**

**Rodger Farley**

**Dave Palace**

**Mick Correia**

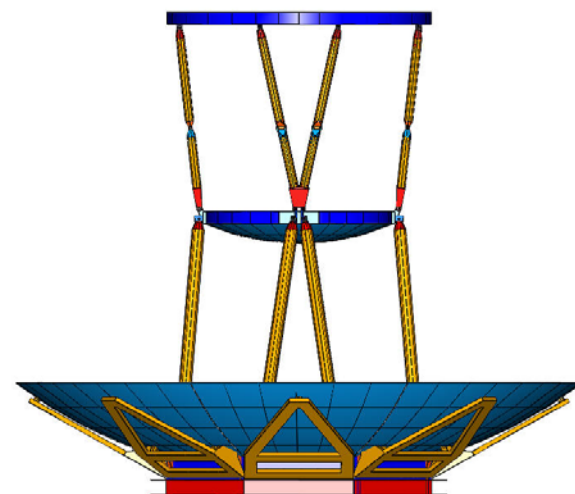
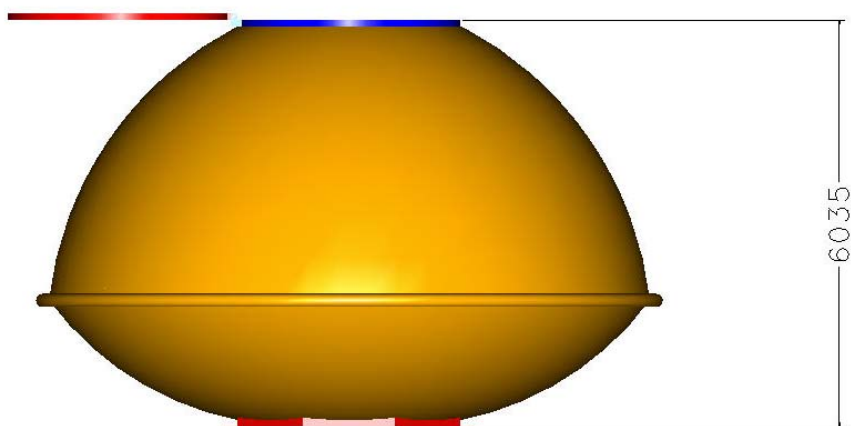
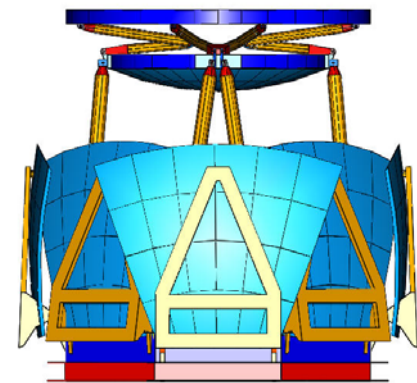
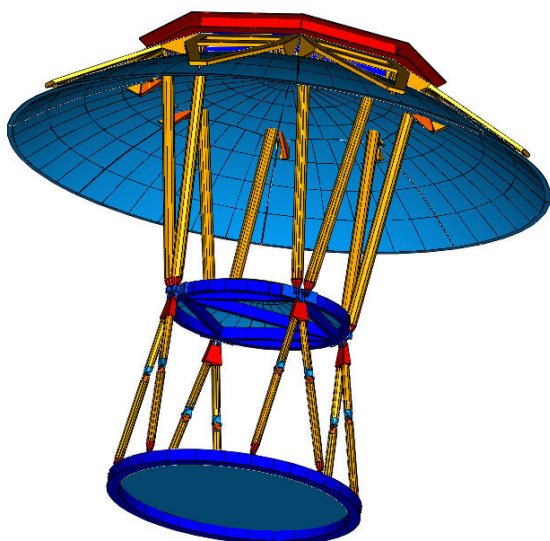
**18 January 2002**



NASA GODDARD SPACE FLIGHT CENTER

# CAD Views of OWL

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# Major Items

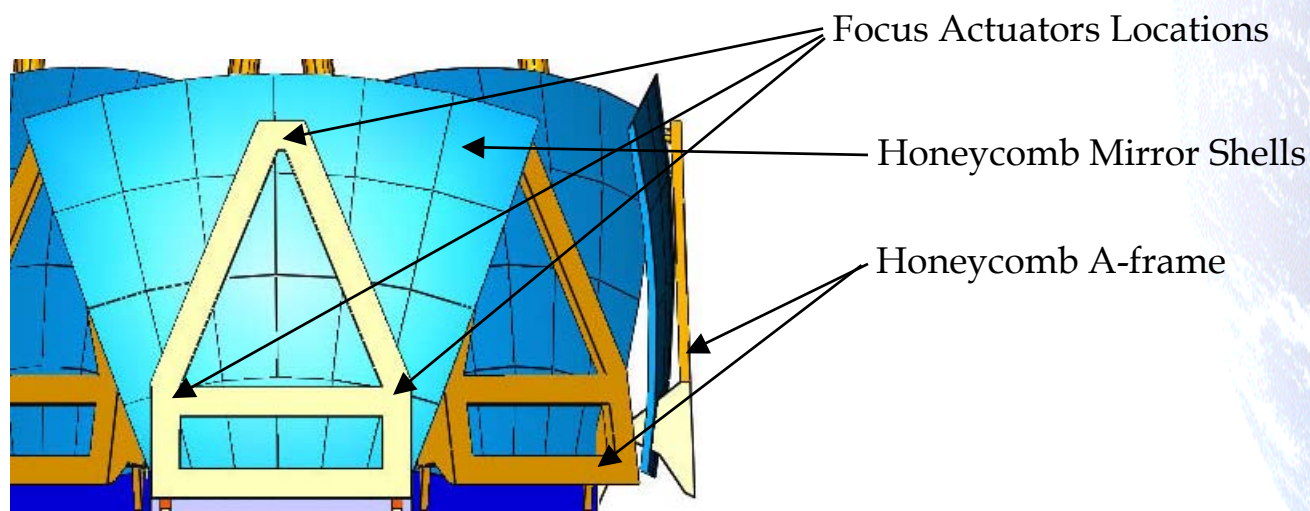
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- 8 deployable mirror petals + 1 fixed center shell
- A-frame support structures behind each deployed mirror petals
- Tip-tilt-focus adjustment actuators, 3 per deployed mirror petals
- Octagonal ring base structure, I/F to spacecraft bus
- Fixed detector plane support structure
- Deployable corrector plate support structure
- Inflatable light-tight, thermal-debris shield
- Articulating light-tight aperture cover ("lens cap")
- Launch-lock and release mechanisms for petals, corrector, and lens cap
- Motorized deployments for petals, corrector
- Most materials will be low CTE composite graphite-cyanate ester

# Primary Mirror

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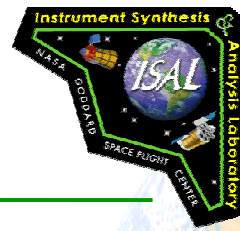
- Mirrored shells are 40 to 50 mm graphite composite honeycomb
- 3-point mounted to the honeycomb A-frame back-up structure
- Each of the 3 points has a ball-screw actuator connected via flexures
- Inner hinge point A-frame design
- Outer offset-hinge A-frame design





# Detector plane and Corrector plate

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Domed corrector in a geodesic mosaic structure for enormous strength with the least obscuration

Deployable corrector plate folding legs

Motorized hinge

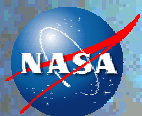
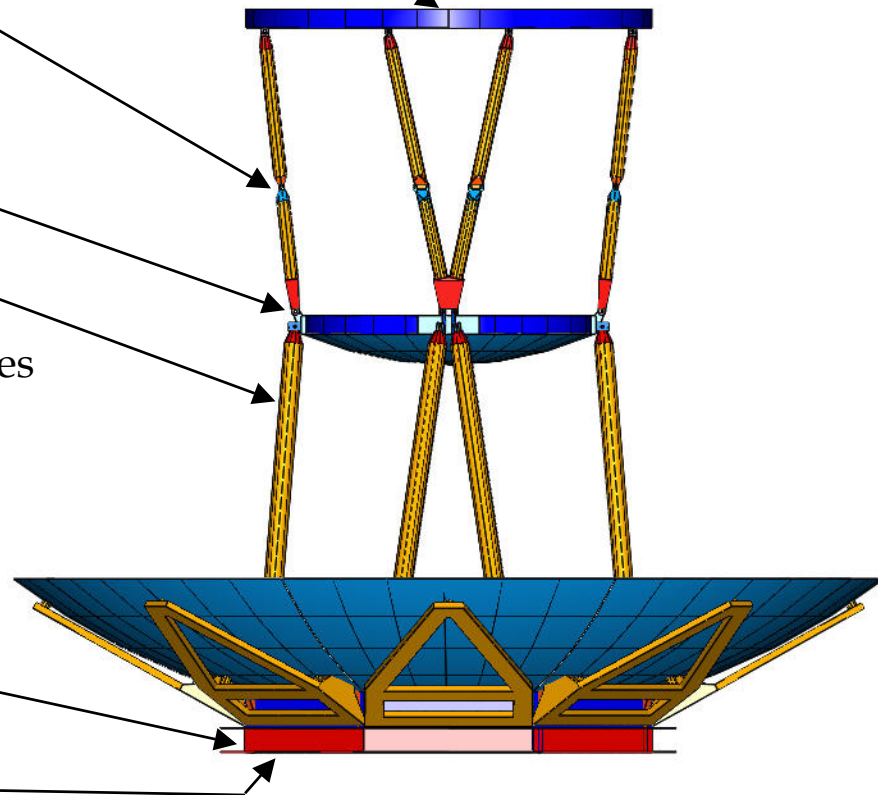
Fixed Detector Plane Supports

150mm diameter, 2mm wall tubes

Composite graphite

Inflatable light-tight debris shield containment box

S/C interface



# Deployment Systems

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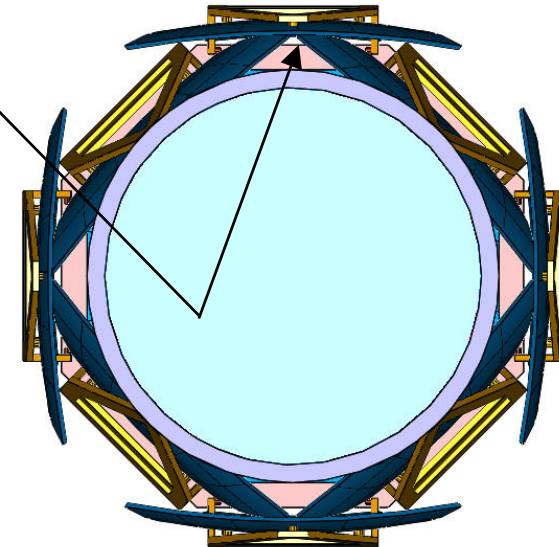
Mirror shells “hold hands” for launch.  
More actuators will be necessary for  
good load path and high natural  
vibration frequency

A-frames motor out sequentially

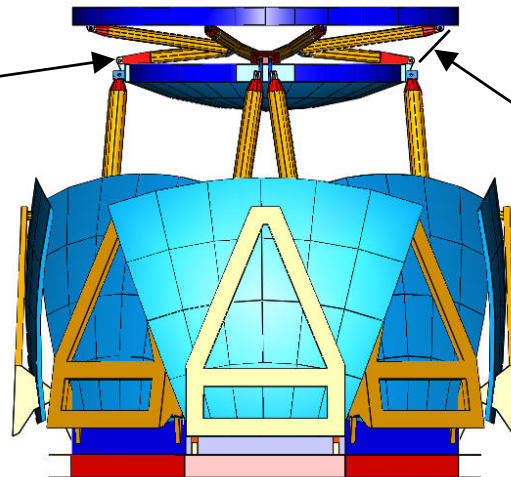
Latch motors pull A-frame into  
preloaded stops

Motorized hinge raises  
the corrector plate

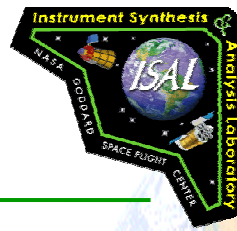
3-way HOP launch lock, 4 places



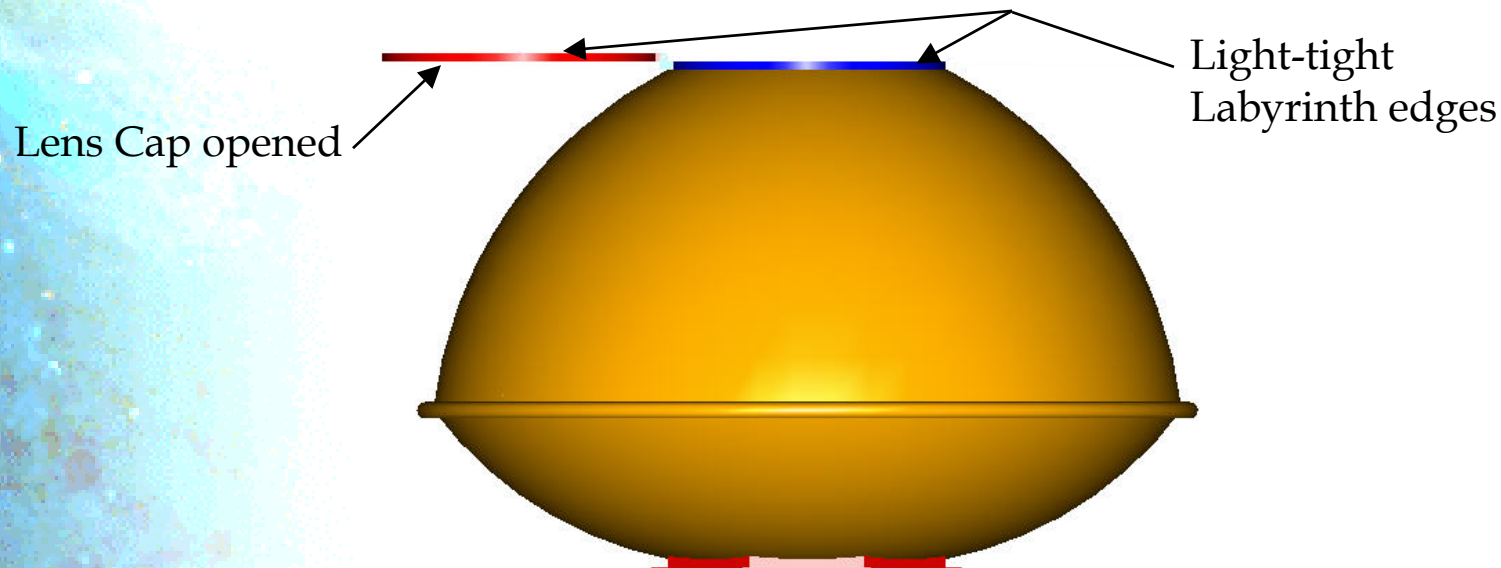
Launch lock  
stanchion not  
shown, 4 places



# Inflatable light-tight debris shield

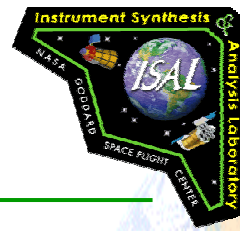


- Made of 10 layer MLI plus “bladder” plus micrometeoroid shield
- Shaped as a “jiffy-pop”, with rigidized toroid and 8 ribs
- Mass of MLI ~ 60 kg
- Mass of micrometeoroid material ~ 50 kg
- Mass of titanium pressure tank ~ 11 kg, 300mm diameter, 9mm thick wall
- Total mass with plumbing, valves, and container box ~180 kg



# Some Numbers

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- 24 focus adjustment micron level actuators
- 8 petal deployment motors
- 8 petal latch motors
- 4 corrector plate deployment motors
- 1 lens cap motor
- 1 lens cap launch lock/reseal motor
- 8-10 inflation valve motor/solenoids
- 8 High Output Paraffin actuators for the inflated shield container
- 4-16 HOPs for the petal launch lock-release mechanisms
- 4 HOPs for the corrector plate launch lock-release mechanisms
- **AND THEN THERE IS REDUNDANCY TO CONSIDER!!**
- The Instrument Module ~ 2000+ kg
- Launch mass to 0 deg inclination 15000 km orbit is ~7000 kg

